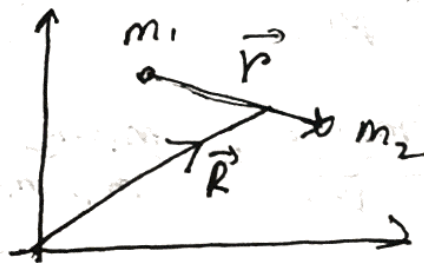


The Central Force Problem



$$\vec{R} = \frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2}, \quad \vec{r} = \vec{r}_2 - \vec{r}_1$$

$$\vec{r}_i' = \vec{r}_i - \vec{R}$$



We saw that

$$T = \frac{1}{2} \sum_i m_i \vec{v}_i^2 = \frac{1}{2} M \vec{V}^2 + \frac{1}{2} \sum_i m_i \vec{v}_i'^2$$

$$M = \sum_i m_i \quad \vec{V} = \dot{\vec{R}} \quad \vec{v}_i' = \dot{\vec{r}}_i'$$

$$\vec{r}_1' = \vec{r}_1 - \frac{m_1 \vec{r}_1 + m_2 \vec{r}_2}{m_1 + m_2} = \frac{(m_1 + m_2) \vec{r}_1 - (m_1 \vec{r}_1 + m_2 \vec{r}_2)}{m_1 + m_2}$$

$$\boxed{\vec{r}_1'} = \frac{m_2 (\vec{r}_1 - \vec{r}_2)}{m_1 + m_2} = - \frac{m_2}{m_1 + m_2} \vec{r} = - \frac{m_2}{M} \vec{r}$$

$$\vec{r}_2' = \frac{m_1 (\vec{r}_2 - \vec{r}_1)}{m_1 + m_2} = \frac{m_1}{m_1 + m_2} \vec{r} = \frac{m_1}{M} \vec{r}$$

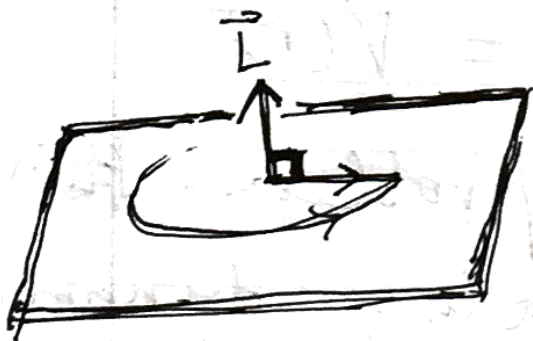
Conserved quantities and first integrals.

$$L = T - V = \frac{1}{2} m \dot{\vec{r}}^2 - V(r)$$

Symmetries: Rotational invariance
Time-translation "

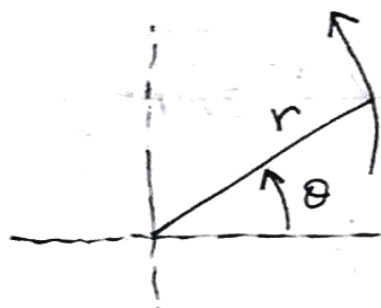
→ Angular momentum conservation
Energy "

$\vec{L} = \vec{r} \times \vec{p}$ is conserved

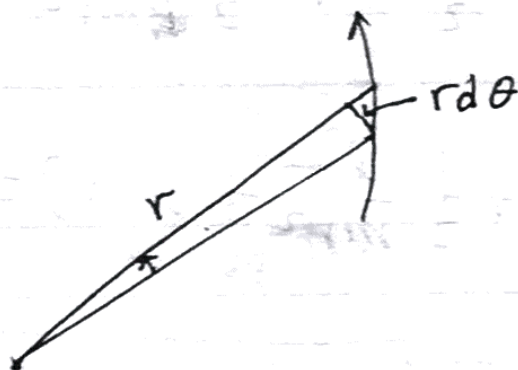


$\vec{r}$ is always $\perp$ to $\vec{L}$
So $\vec{r}$ lies in a plane.

Set up polar coordinates.



$$L = \frac{1}{2} m (\dot{r}^2 + r^2 \dot{\theta}^2) - V(r)$$



$$\vec{r} \times \vec{p} dt = m \vec{r} \times \vec{v} dt = m \vec{r} \times d\vec{r}$$



$$dA = \frac{1}{2} r^2 d\theta$$

$$\frac{dA}{dt} = \frac{|\vec{L}|}{2m} = \text{Const.}$$

Anyway $p_{\theta} = m r^2 \dot{\theta} = l = |\vec{L}|$

Other equation

$$\begin{aligned} \frac{d}{dt}(m \dot{r}) &= m r \dot{\theta}^2 - \frac{dV}{dr}(r) \\ &= \frac{l^2}{m r^3} - \frac{dV}{dr}(r) \end{aligned}$$

using $\dot{\theta} = \frac{l}{m r^2}$

$$\frac{d}{dt} m \dot{r} = - \frac{d}{dr} \left(V(r) + \frac{l^2}{2mr^2} \right)$$

Multiply by $\dot{r}$

$$\frac{d}{dt} \left(\frac{m \dot{r}^2}{2} \right) = - \frac{d}{dt} \left(V(r) + \frac{l^2}{2mr^2} \right)$$

$$\frac{d}{dt} \left[\frac{m \dot{r}^2}{2} + \frac{1}{2} \frac{l^2}{mr^2} + V(r) \right] = 0$$

$$\frac{m \dot{r}^2}{2} + \frac{1}{2} \frac{l^2}{mr^2} + V(r) = E = \text{constant.}$$

$$\dot{r} = \sqrt{\frac{2}{m} \left(E - V(r) - \frac{l^2}{2mr^2} \right)}$$

$$t = \int_{r_0}^r \frac{dr}{\sqrt{\frac{2}{m} \left(E - V(r) - \frac{l^2}{2mr^2} \right)}}$$

$$t = t(r)$$

$$\Rightarrow r = r(t)$$

$$\dot{\theta} = \frac{l}{m r^2(t)}$$

$$\theta = l \int_0^t \frac{dt'}{m r^2(t')} + \theta_0$$

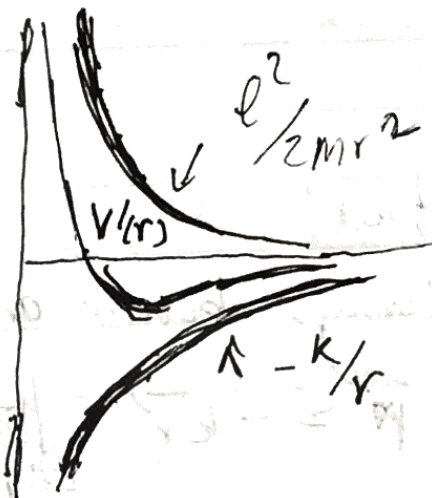
■ Equivalent one-dim problem

$$E = \frac{1}{2} m \dot{r}^2 + V'(r)$$

$$V'(r) = V(r) + \frac{l^2}{2mr^2}$$

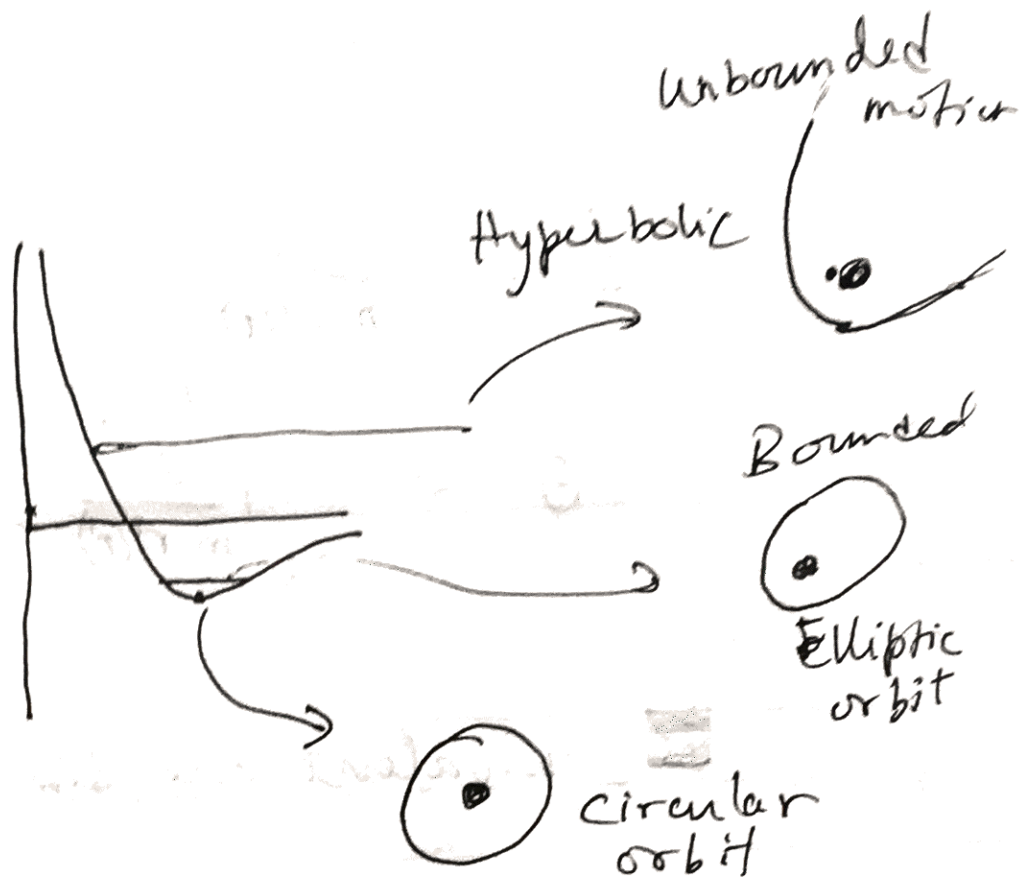
Consider the special case

$$V(r) = -\frac{k}{r}$$



Large r $-\frac{k}{r}$
dominant

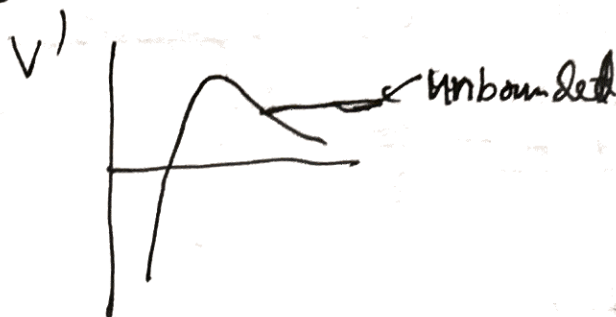
Small r $\frac{l^2}{2mr^2}$
dominant



$V(r)$ isn't slower than $1/r^2$ ^{stable large} bounded orbits are hard to are hard to get.

Other potentials

Ugly: $V(r) = -\frac{a}{r^3}$



Nice!

$V(r) = \frac{1}{2}kr^2$



always bound orbits

$\vec{F} = -k\vec{r}$ ~~Elliptic~~ Elliptic orbits



Virial Theorem:

More general than central motion, Gives information about statistical averages

Define $G = \sum_i \vec{p}_i \cdot \vec{r}_i$

$$\begin{aligned} \dot{G} &= \sum_i \dot{\vec{r}}_i \cdot \vec{p}_i + \sum_i \dot{\vec{p}}_i \cdot \vec{r}_i \\ &= \sum_i m \vec{v}_i^2 + \sum_i \vec{F}_i \cdot \vec{r}_i \\ &= 2T + \sum_i \vec{F}_i \cdot \vec{r}_i \end{aligned}$$

Integrating from $t=0$ to $t=\tau$ and dividing by τ .

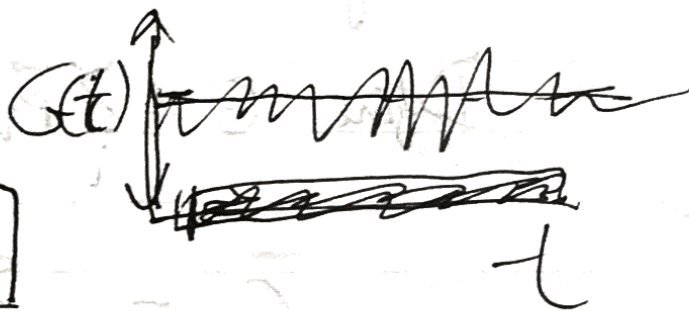
$$\frac{1}{\tau} [G(\tau) - G(0)] = 2\bar{T} + \sum_i \bar{\vec{F}_i \cdot \vec{r}_i}$$

Where ~~$\overline{\mathcal{O}(t)}$~~ $\frac{\int_0^\tau \mathcal{O}(t) dt}{\int_0^\tau dt}$ is the time average $\bar{\mathcal{O}}$

If the motion is statistically periodic

$$\frac{G(\tau) - G(0)}{\tau} \rightarrow 0$$

as $\tau \rightarrow \infty$



$$\Rightarrow \boxed{\bar{T} = -\frac{1}{2} \sum_i \vec{F}_i \cdot \vec{r}_i}$$

Application to ideal gas:

$$\bar{T} = -\frac{1}{2} \sum_i \vec{F}_i \cdot \vec{r}_i$$



$$\frac{3}{2} N k_B T = \frac{1}{2} \int \vec{r} \cdot (P \hat{n} dA)$$

$$= \frac{1}{2} P \int \vec{r} \cdot d\vec{A}$$

$$= \frac{1}{2} P \int \vec{\nabla} \cdot \vec{r} dV$$

$$= \frac{3}{2} P V$$

$$PV = Nk_B T$$

In principle, with weak collisions
one could 'correct' the equation
of state for non-ideal gases

For conservative systems

$$F_i = \nabla_i V$$

$$\bar{T} = \frac{1}{2} \sum_i \vec{r}_i \cdot \vec{\nabla}_i V$$

Aside on homogeneous functions

$$f(x_1, \dots, x_n) = \lambda^k f(\lambda x_1, \dots, \lambda x_n)$$

Degree k homogeneous functions

Put $\lambda = 1 + \epsilon$ ϵ small

$$f(x_1 + \epsilon x_1, x_2 + \epsilon x_2, \dots, x_N + \epsilon x_N)$$

$$= (1 + \epsilon)^k f(x_1, \dots, x_N)$$

Expanding up to first order in ϵ
we get

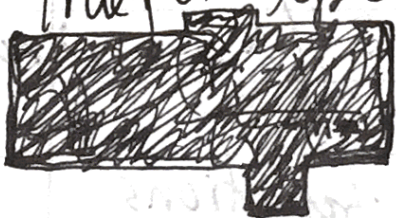
$$\sum_i x_i \frac{\partial f}{\partial x_i} = k f, \text{ Euler's Theorem}$$

$$\text{So, if } V(\lambda \vec{r}_1, \dots, \lambda \vec{r}_N) = \lambda^{n+1} V(\quad)$$

$$[\text{Force goes as } \lambda^n]$$

$$\overline{T} = \frac{1}{2} \sum_i \vec{r}_i \cdot \nabla_i V = \frac{n+1}{2} \overline{V}$$

True for systems with central forces



$$V_{ij}(r_{ij}) \sim r_{ij}^{n+1}$$

$$F_{ij} \sim r_{ij}^n$$

When we have only one pair or many pairs, this rule applies

For inverse square law $n = -2$

$$\overline{T} = -\frac{1}{2} \overline{V}$$

Harmonic oscillator system $n = 1$

$$\overline{T} = \overline{V}$$

$$\overline{T} = -\frac{1}{2} \overline{V}$$

↓
Dark matter
Suspicion

$$\overline{T} = \overline{V}$$

↓
Equipartition
of energy for
molecular
vibrations.

Notice

$$\vec{r}_i \rightarrow \lambda \vec{r}_i$$

$$t \rightarrow \lambda^{1+s} t$$

$$T \rightarrow \lambda^{1-s} T$$

$$\ddot{\vec{r}}_i = \vec{F}_i$$

$$\lambda^{1+s} \ddot{\vec{r}}_i = \lambda^n \vec{F}_i$$

$$\lambda^{1+s} \ddot{\vec{r}}_i = \lambda^n \vec{F}_i$$

Back to central potential

$$\dot{\theta} = \frac{l}{mr^2}$$

$$\frac{d}{dt} = \dot{\theta} \frac{d}{d\theta} = \frac{l}{mr^2} \frac{d}{d\theta}$$

$$m\ddot{r} - \frac{l^2}{mr^3} = f(r)$$

$$\frac{1}{r^2} \frac{d}{d\theta} \left(\frac{l}{mr^2} \frac{dr}{d\theta} \right) - \frac{l^2}{mr^3} = f(r)$$

$$\frac{d}{d\theta} \left[\frac{1}{r^2} \frac{dr}{d\theta} \right] - \frac{1}{r} = \frac{mr^2 f(r)}{l^2}$$

$$-\frac{d}{d\theta} \frac{du}{d\theta} - u = \frac{m}{l^2} \frac{d}{du} V\left(\frac{1}{u}\right)$$

$$\frac{d^2 u}{d\theta^2} + u = -\frac{m}{l^2} \frac{d}{du} V\left(\frac{1}{u}\right)$$

Example: Kepler problem

$$V(r) = -\frac{k}{r}$$

$$V(u) = -ku$$

$$\frac{d^2 u}{d\theta^2} + u = \frac{m}{\ell^2} \frac{d(ku)}{du}$$

$$\frac{d^2 u}{d\theta^2} + \left(u - \frac{mk}{\ell^2}\right) = 0$$



$$u = \frac{mk}{\ell^2} + C \cos(\theta - \theta_0)$$

$$= \frac{mk}{\ell^2} (1 + e \cos(\theta - \theta_0))$$

$$e = \sqrt{1 + \frac{2E\ell^2}{mk^2}}$$

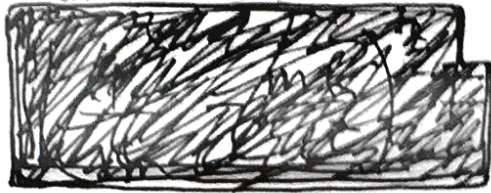
from the following
considerations:

$$E = \frac{1}{2} m \dot{r}^2 + \frac{\ell^2}{2mr^2} - \frac{k}{r}$$

At max or min of r , $\dot{r} = 0$

$$\frac{1}{r_m} = \frac{mk}{l^2}(1 \pm e) \quad \dots$$

r_m could correspond to max or min.



$$\left(\frac{l^2}{mk r_m} - 1 \right)^2 = e^2$$

$$1 - \frac{2l^2}{mk r_m} + \frac{l^4}{m^2 k^2 r_m^2} = e^2$$

$$1 + \frac{2l^2}{mk^2} \left(-\frac{k}{r_m} + \frac{l^2}{2m r_m^2} \right) = e^2$$

E

$$\text{So } 1 + \frac{2El^2}{mk^2} = e^2$$

Alternatively we could have done the following:

$$\begin{aligned} \theta - \theta_0 &= l \int_0^t \frac{dt'}{mr^2(t')} = l \int_0^t \frac{dr}{mr^2 \dot{r}} \\ &= l \int_{r_0}^r \frac{dr}{r^2 \sqrt{2(E - V(r))}} \end{aligned}$$

$$= \frac{l}{\sqrt{2m}} \int_{r_0}^r \frac{dr'}{r'^2 \sqrt{E - \frac{l^2}{2mr'^2} - V(r')}}$$

Using $u = \frac{1}{r}$

$$= -\frac{l}{\sqrt{2m}} \int_{u_1}^u \frac{du'}{\sqrt{E - \frac{l^2}{2m} u'^2 - V\left(\frac{1}{u'}\right)}}$$

For the Kepler problem $V\left(\frac{1}{u}\right) = -ku$

$$\begin{aligned} & \sqrt{E + ku - \frac{l^2}{2m} u^2} \\ &= \sqrt{E + \frac{mk^2}{2l^2} - \frac{l^2}{2m} \left(u - \frac{mk}{l^2}\right)^2} \\ &= \frac{l}{\sqrt{2m}} \sqrt{\left(1 + \frac{2El^2}{mk^2}\right) \frac{mk^2}{l^4} - \left(u - \frac{mk}{l^2}\right)^2} \end{aligned}$$



Appropriate transformation

$$u - \frac{mk}{l^2} = \frac{mk}{l^2} e \cos \phi$$

With

$$e = \sqrt{1 + \frac{2El^2}{mk^2}}$$

$$\theta - \theta_0 = \int_0^{\phi} \frac{-\frac{mk}{l^2} e \sin \phi \, d\phi}{\sqrt{\frac{m^2 k^2}{l^4} e^2 - \frac{m^2 k^2}{l^4} e^2 \cos^2 \phi}}$$

$$= \phi$$

$$\theta \leftrightarrow \phi$$

$$u - \frac{mk}{l^2} = \frac{mk}{l^2} e \cos(\theta - \theta_0)$$

$$\frac{1}{r} = u = \frac{mk}{l^2} \left[1 + e \cos(\theta - \theta_0) \right]$$

$$r = \frac{l^2 / mk}{1 + e \cos(\theta - \theta_0)}$$

Classification of orbits

$e > 1$	$E > 0$	hyperbola
$e = 1$	$E = 0$	Parabola
$e < 1$	$E < 0$	ellipse
$e = 0$	$E = -\frac{mk^2}{2l^2}$	circle

averaging $E = T + V$

$$E = \overline{T} + \overline{V} = -\frac{\overline{V}}{2} + \overline{V} = \frac{\overline{V}}{2}$$

$$E = -\frac{k}{2r} \Rightarrow \frac{k}{2} \overline{r^{-1}}$$

If $r = r_0$ (circle), balance between 'centrifugal' force' and attraction

$$\frac{k}{r_0^2} = \frac{mv^2}{r_0} = \frac{l^2}{mr_0^3}$$

$$\Rightarrow r_0 = \frac{l^2}{mk}$$

$$\Rightarrow E = -\frac{k}{2} \frac{1}{r_0} = -\frac{mk^2}{2l^2}$$

$$e = 0$$

Elliptic orbit



$$E - \frac{l^2}{2mr^2} + \frac{k}{r} = 0$$

$$r^2 + \frac{k}{E} r - \frac{l^2}{2mE} = 0$$

$$r_1 + r_2 = -\frac{k}{E}$$

$$a = \frac{r_1 + r_2}{2} = -\frac{k}{2E}$$

↑
Semimajor axis

Note that

$$r_{1,2} = a(1 \pm e)$$

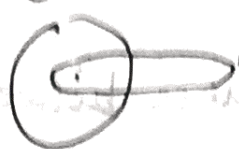
$$r_1 r_2 = a^2(1 - e^2) = -\frac{l^2}{2mk} \frac{k}{E} = \frac{l^2}{mkE}$$

$$1 - e^2 = \frac{l^2}{mka} \Rightarrow e = \sqrt{1 - \frac{l^2}{mka}}$$

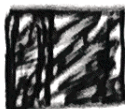
Also, using $a(1 - e^2) = \frac{l^2}{mk}$

$$r = \frac{a(1 - e^2)}{1 + e \cos(\theta - \theta_0)}$$

$$e = 0$$



e close to 1.

Time  and angle/radius

$$t = \int_{r_0}^r \frac{dr}{\dot{r}} = \frac{m}{\sqrt{2}} \int_{r_0}^r \frac{dr}{\sqrt{mr^2/2}} = \frac{\sqrt{m}}{\sqrt{2}} \int_{r_0}^r \frac{dr}{\sqrt{\frac{k}{r} - \frac{l^2}{2mr^2} + e}}$$

Also using $dt = \frac{mr^2}{l} d\theta$

$$t = \int_{\theta_0}^{\theta} \frac{mr^2}{l} d\theta = \frac{l^3}{mk^2} \int_{\theta_0}^{\theta} \frac{d\theta}{(1 + e \cos(\theta - \theta_0))^2}$$

Sometimes it is better to express things in terms of eccentric anomaly

$$r = a(1 - e \cos \psi)$$

$$\sqrt{\frac{m}{2}} \int \frac{dr}{\sqrt{\frac{k}{r} - \frac{l^2}{2mr^2} + E}} = \sqrt{\frac{m}{2}} \int \frac{r dr}{\sqrt{Er^2 + kr - \frac{l^2}{2m}}}$$

$$= \sqrt{\frac{m}{2}} \int \frac{r dr}{\sqrt{E \left(r^2 + \frac{k}{E} r - \frac{l^2}{2mE} \right)}}$$

$\uparrow$ $-2a$ $\uparrow$ $a^2(1-e^2)$

$$E = -\frac{k}{2a}$$

$$= \sqrt{\frac{m}{2k}} \int \frac{r dr}{\sqrt{-\frac{1}{2a} (r^2 - 2ar + a^2(1-e^2))}} = \sqrt{\frac{ma}{k}} \int \frac{r dr}{\sqrt{a^2 e^2 - (r-a)^2}}$$

$$r = a(1 - e \cos \psi)$$

$$t = \sqrt{\frac{ma}{k}} \int \frac{a^2(1 - e \cos \psi) \sin \psi d\psi}{\sqrt{a^2 e^2 (1 - \cos^2 \psi)}} = \sqrt{\frac{ma^3}{k}} \int_0^\psi (1 - e \cos \psi) d\psi$$

Note that $T = 2\pi \sqrt{\frac{ma^3}{k}}$ from integrating
 around the period orbit.

Laplace-Runge-Lenz Vector

$$\vec{p} = f(r) \frac{\vec{r}}{r}$$

$$\vec{p} \times \vec{L} = m f(r) \frac{\vec{r}}{r} \times (\vec{r} \times \dot{\vec{r}})$$

$$= \frac{m f(r)}{r} [\vec{r} (\vec{r} \cdot \dot{\vec{r}}) - r^2 \dot{\vec{r}}]$$

$$= \frac{m f(r)}{r} [\vec{r} r \dot{r} - r^2 \dot{\vec{r}}]$$

$$\vec{L} \text{ is constant, so } \frac{d}{dt}(\vec{p} \times \vec{L}) = -m f(r) r \frac{d}{dt} \left(\frac{\vec{r}}{r} \right)$$

For the Kepler problem $f(r) = -\frac{k}{r^2}$

$$\frac{d}{dt} \left(\vec{p} \times \vec{L} - \frac{mk\vec{r}}{r} \right) = 0$$

$$\frac{d}{dt} \vec{A} = 0$$



$$\vec{A} = \vec{p} \times \vec{L} - \frac{mk\vec{r}}{r}$$

$$\begin{aligned} \vec{A} \cdot \vec{r} &= \vec{r} \cdot (\vec{p} \times \vec{L}) - mkr \\ &= \vec{L} \cdot (\vec{r} \times \vec{p}) - mkr \\ &= L^2 - mkr \end{aligned}$$

$$A \cos \theta = l^2 - mkr$$

$$\frac{1}{r} = \frac{mk}{l^2} \left(1 + \frac{A}{mk} \cos \theta \right)$$



$$A = mke$$

$\vec{A}$ is not completely independent of E & $\vec{L}$.

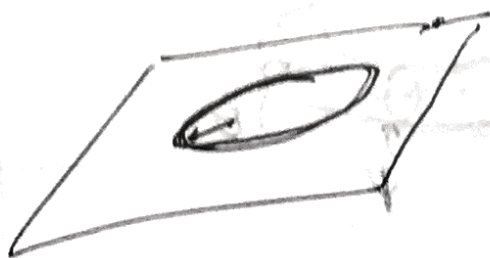
$$A^2 = m^2 k^2 e^2 = m^2 k^2 \left(1 + \frac{2Ee^2}{mk^2} \right)$$

$$= m^2 k^2 + 2mEe^2$$

$E, \vec{L} \rightarrow 4$ conserved quantities

$\vec{A}$ adds one more conserved quantity

$\Rightarrow$ angle of perihelion.

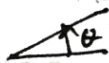


Existence of such extra conserved quantity has something to do with having closed orbits in the Kepler problem.

Consider perturbations of orbits around a circular one.

Focus on 2D

Angle:

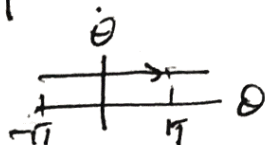


$$\dot{\theta} \approx \frac{l}{mr_0^2} = \omega_\theta$$

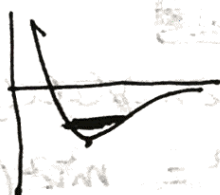


$$\theta \sim \omega_\theta t$$

$$p_\theta = l \text{ constant}$$

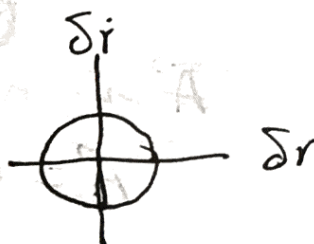


Radial

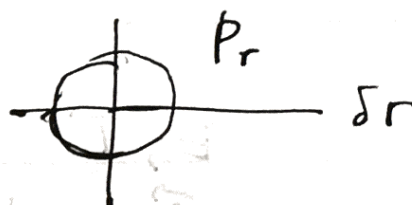


$$\delta r = r - r_0$$

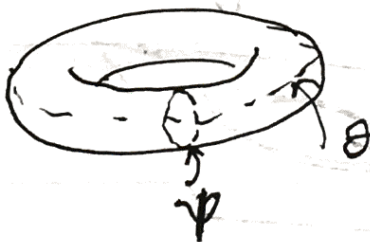
$$\delta r = a e^{i\omega_r \psi(t)}$$



$$E - E_0 = \text{constant}$$



$$\psi(t) \approx \omega_r t$$



In general ~~the~~ ω_θ and ω_r are not commensurate



To have closed orbits for generic initial conditions, First order analysis suggests

$$f(r) = -\frac{k}{r^3 - \beta^2} \text{ with } \beta \text{ rational}$$

Higher order analysis $\Rightarrow \beta^2 = 1, 4$
 Bertrand's Theorem.

Additional conserved quantities
 Foliation of the ~~tota~~ torus



[Circular orbit: $f(r_0) + \frac{l^2}{mr_0^3} = 0$
 $\omega_0 = \frac{l}{mr_0^2}$

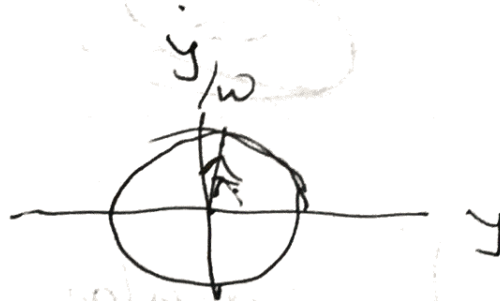
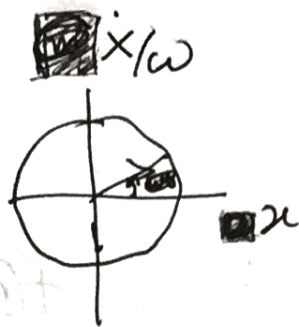
Linearize: $m \delta r = \left(f'(r_0) - \frac{3l^2}{mr_0^4} \right) \delta r$
 $= - \left(3 - \frac{mr_0^4 f'(r_0)}{l^2} \right) \omega_0^2 \delta r$
 $= - \left(3 + \frac{r_0 f'(r_0)}{f(r_0)} \right) \omega_0^2 \delta r$
 $= - \beta^2 \omega_0^2 \delta r$

β const fnc of r_0 and rational $\Rightarrow$ const.

$\frac{r_0 f'(r_0)}{f(r_0)} = \beta^2 - 3 \Rightarrow f(r) = -\frac{k}{r^{3-\beta^2}}$
 with β rational

Additional conserved quantities
for isotropic harmonic
oscillator

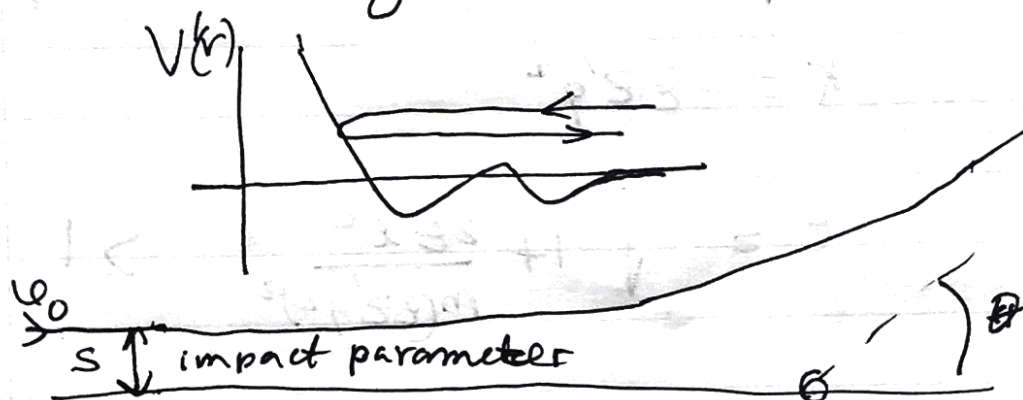
$$\frac{m}{2} \dot{x}_i \dot{x}_j + \frac{m\omega^2}{2} x_i x_j$$



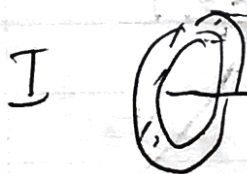
~~Angle~~

Angle between these phase space
vectors are preserved.

Scattering in a central force



$$l = m v_0 s = s \sqrt{2mE}$$



$$2\pi I s ds = 2\pi I \sigma(\theta) \sin\theta d\theta$$

↑
area unit

Express $s = s(\theta, E)$

$$\sigma(\theta) = \frac{s}{\sin\theta} \frac{ds}{d\theta}$$

Application to Coulomb Repulsion

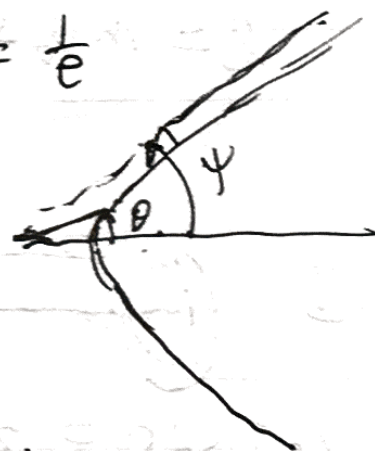
$$k = -zz'q^2$$

$$e = \sqrt{1 + \frac{2El^2}{m(zz'q^2)^2}} > 1$$

$$\frac{1}{r} = \frac{mzz'q^2}{l^2} (e \cos \theta - 1)$$



$$\cos \theta = \frac{1}{e}$$



$$\theta = \pi - 2\phi$$

$$\phi = \frac{\pi}{2} - \frac{\theta}{2}$$

$$\sin \frac{\theta}{2} = \frac{1}{e} \left(\frac{2El^2}{m(zz'q^2)^2} \right)$$

$$1 + \frac{2El^2}{m(zz'q^2)^2} = e^2 = \sec^2 \frac{\theta}{2} = 1 + \tan^2 \frac{\theta}{2}$$

$$S = \frac{zz'e^2}{2E} \cot^2 \frac{\theta}{2}$$

$$\sigma(\theta) = \frac{1}{4} \left(\frac{zz'e^2}{2E} \right)^2 \csc^4 \frac{\theta}{2}$$

Divergence
at $\theta \rightarrow 0$
Long range interaction
Large
back scattering

Rutherford Scattering: