

$$I_E = \frac{2}{5} M_E R_E^2$$

$$L_{tot} = L_{E-rot} + L_{M-orbit} + L_{M-rot}$$

very small (neglect)

constant →
$$L_{tot} = I_E \left(\frac{2\pi}{T_E} \right) + m_M R^2 \left(\frac{2\pi}{T_{M-orb}} \right)$$

$$K_{E-rot} = \frac{1}{2} I_E \omega_E^2 = \frac{1}{2} I_E \left(\frac{2\pi}{T_E} \right)^2$$

solid Earth – rotates under ocean tidal bulge
- frictional energy loss

$$\Delta E_E = \Delta K_{E-rot} = -W_{tidal-friction}$$

Earth's rotation slows with time
 $T_E(\text{later}) > T_E(\text{earlier})$

$$-W_{tidal} = \Delta K_{E-rot} = \left[\frac{(2\pi)^2}{2} I_E \right] \left(\frac{1}{T_E(\text{later})^2} - \frac{1}{T_E(\text{earlier})^2} \right)$$

constant →
$$L_{tot} = I_E \left(\frac{2\pi}{T_E} \right) + m_M R^2 \left(\frac{2\pi}{T_{M-orb}} \right)$$

decreases increases

L transferred to moon

moon moves away with time

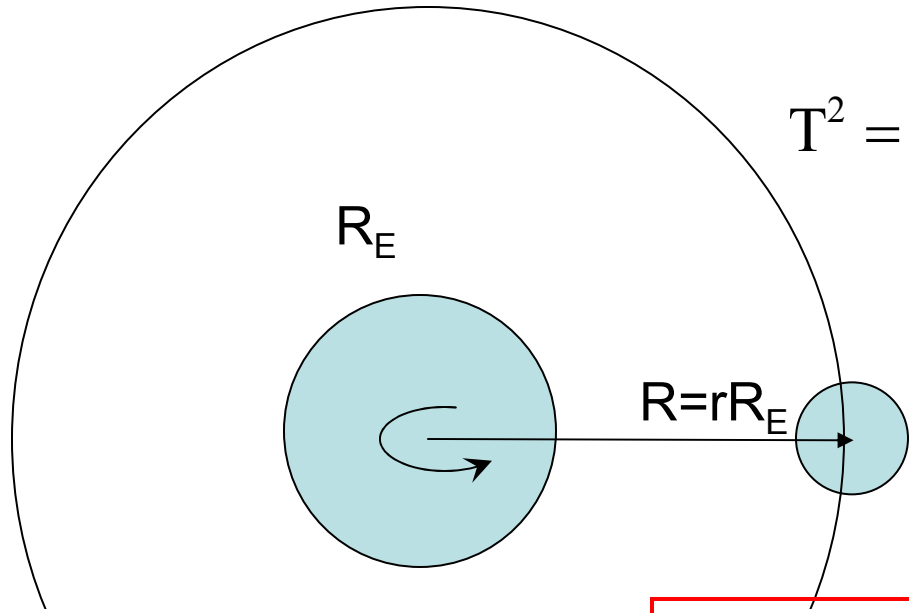
$$L_{tot} = I_E \left(\frac{2\pi}{T_E} \right) + m_M \sqrt{GM_E R} R^{1/2}$$

$$\frac{m^3}{s^2 kg} = G$$

$$T^2 = \frac{(2\pi)^2}{GM_E} R^3$$

$$T^{-2} = \frac{GM_E}{(2\pi)^2 R^3}$$

$$\frac{1}{T} = \frac{\sqrt{GM_E}}{(2\pi)} \frac{1}{R^{3/2}}$$



$$T^2 = \frac{(2\pi)^2 R^3}{GM_E}$$

$$T_{M\text{-orbit}} = 27 \text{ days} = T_{E\text{-rot}27}(\text{now})$$

$$T_{M\text{-orbit}} = \tau T_{E\text{-rot}}(\text{now})$$

$$R = rR_E$$

$$\tau^2 = \left[\frac{(2\pi)^2 R_E^3}{GM_E T_E^2} \right] r^3$$

$$\tau^2 = [0.003584] r^3$$

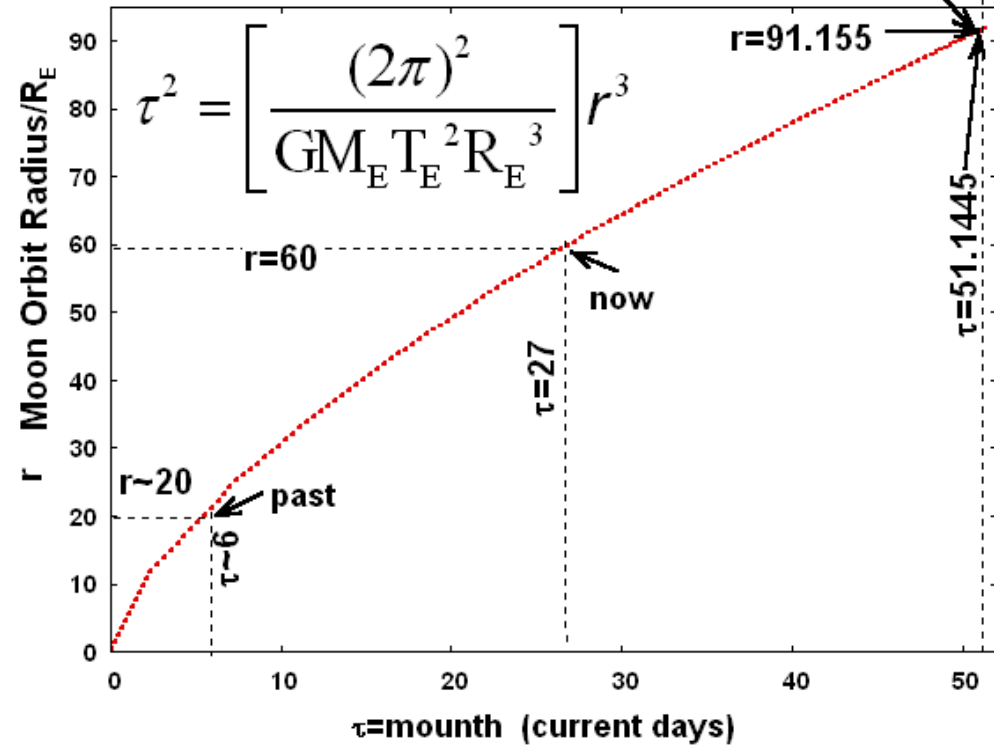
Earth rotation period
= Moon orbit period

distant future

moon-orbit at 91 R_E

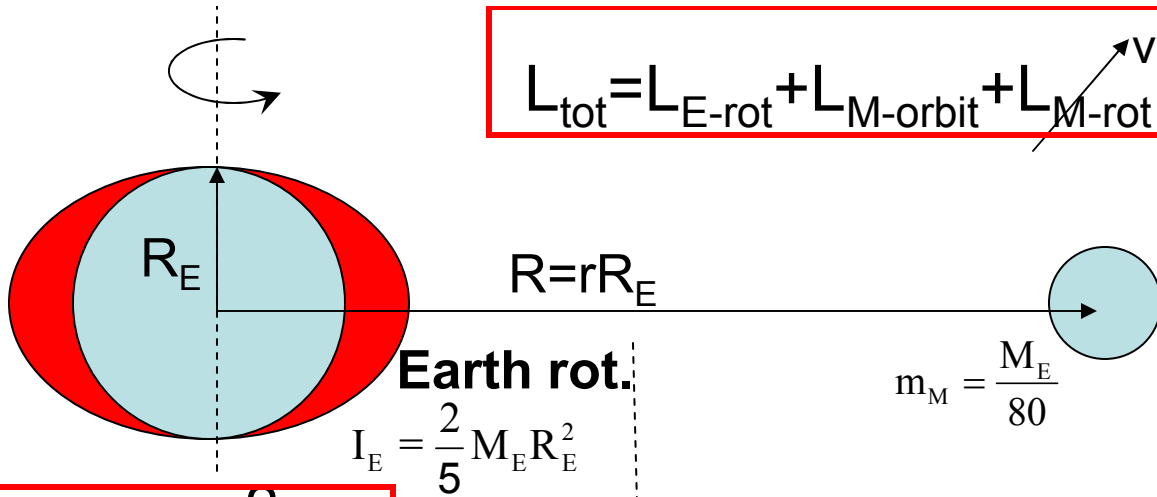
day'=month'=51 days (current)

$$\tau^2 = [0.0033375] r^3$$



distant past

**moon- much closer
day shorter**



$$L_{\text{tot}} = L_{E\text{-rot}} + L_{M\text{-orbit}} + L_{M\text{-rot}} \quad \text{very small (neglect)}$$

$$L_{E\text{-rot}} = \{I\} \frac{2\pi}{T_E}$$

$$L_{E\text{-rot}} = \frac{2}{5} \left[\frac{M_E R_E^2 2\pi}{T_E} \right]$$

$$T_{E\text{-rot}} (\text{at other time}) = t T_{E\text{-rot}} (\text{now})$$

$$L_{E\text{-rot}} = \frac{2}{5t} \left[\frac{M_E R_E^2 2\pi}{T_E} \right]$$

$$L_{\text{totM-orbit}} = L_{E\text{-rot}} + L_{M\text{-orbit}}$$

$$L_{\text{totM-orbit}} = \left\{ \frac{2}{5t} + \frac{r^2}{80\tau} \right\} \left[\frac{M_E R_E^2 2\pi}{T_E} \right]$$

Moon orbit

$$T_{M\text{-orbit}} = 27 \text{ days} = T_{E\text{-rot}27} (\text{now})$$

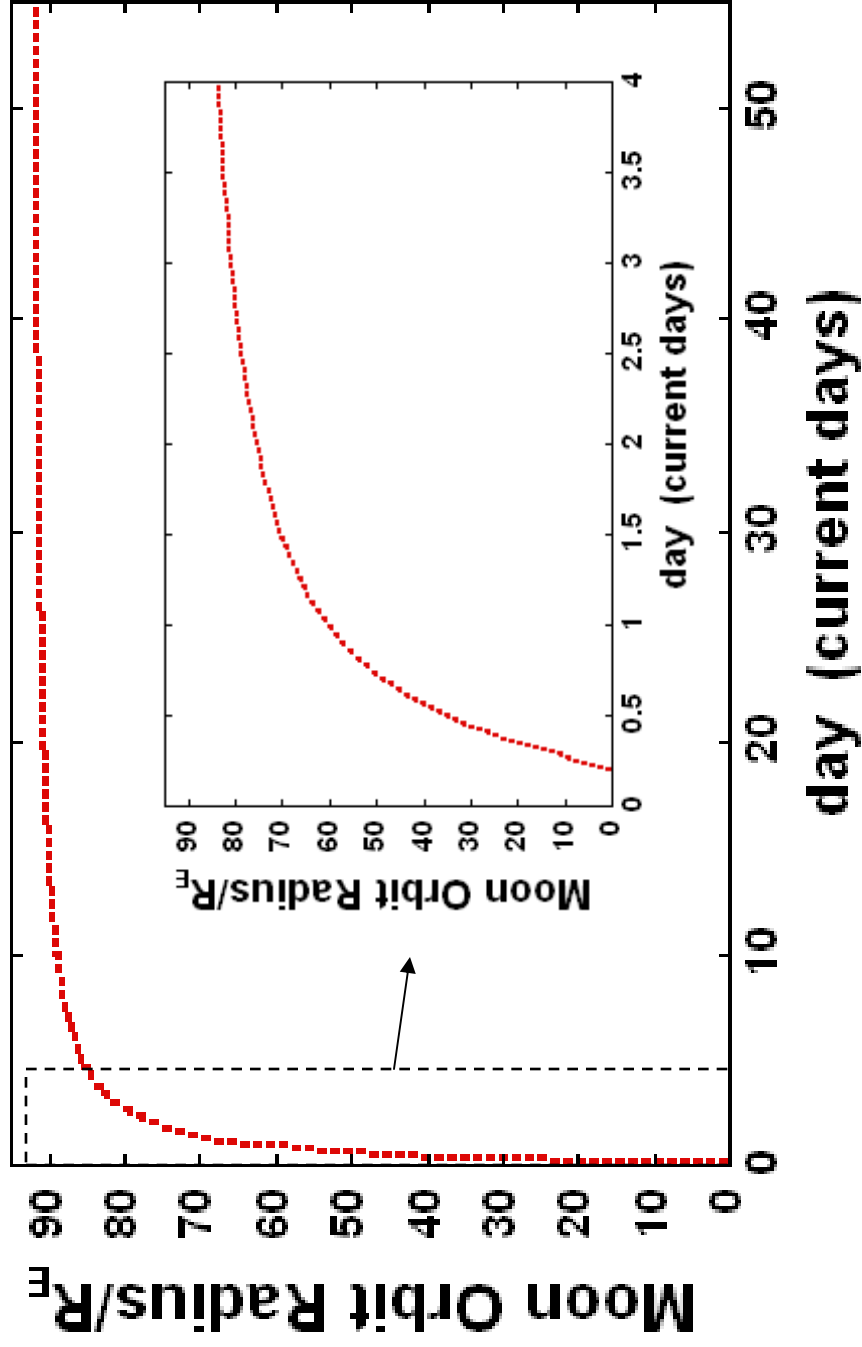
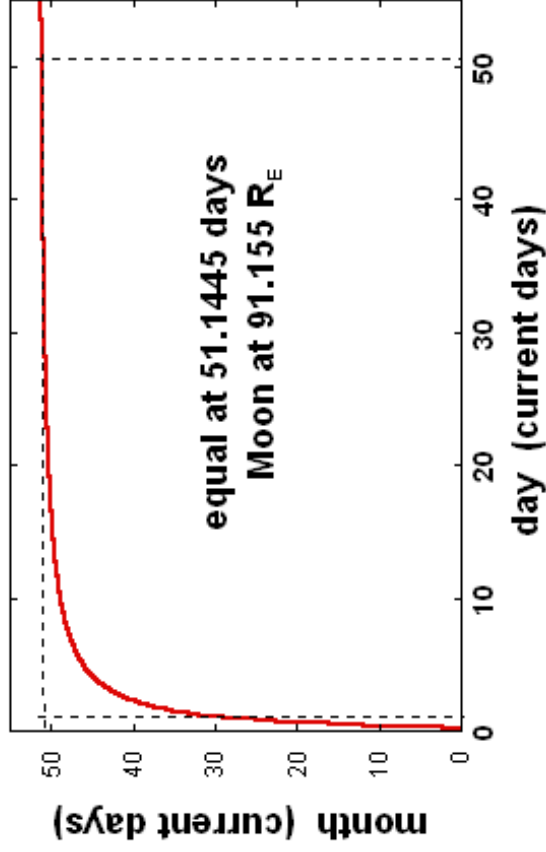
$$T_{M\text{-orbit}} = \tau T_{E\text{-rot}} (\text{now})$$

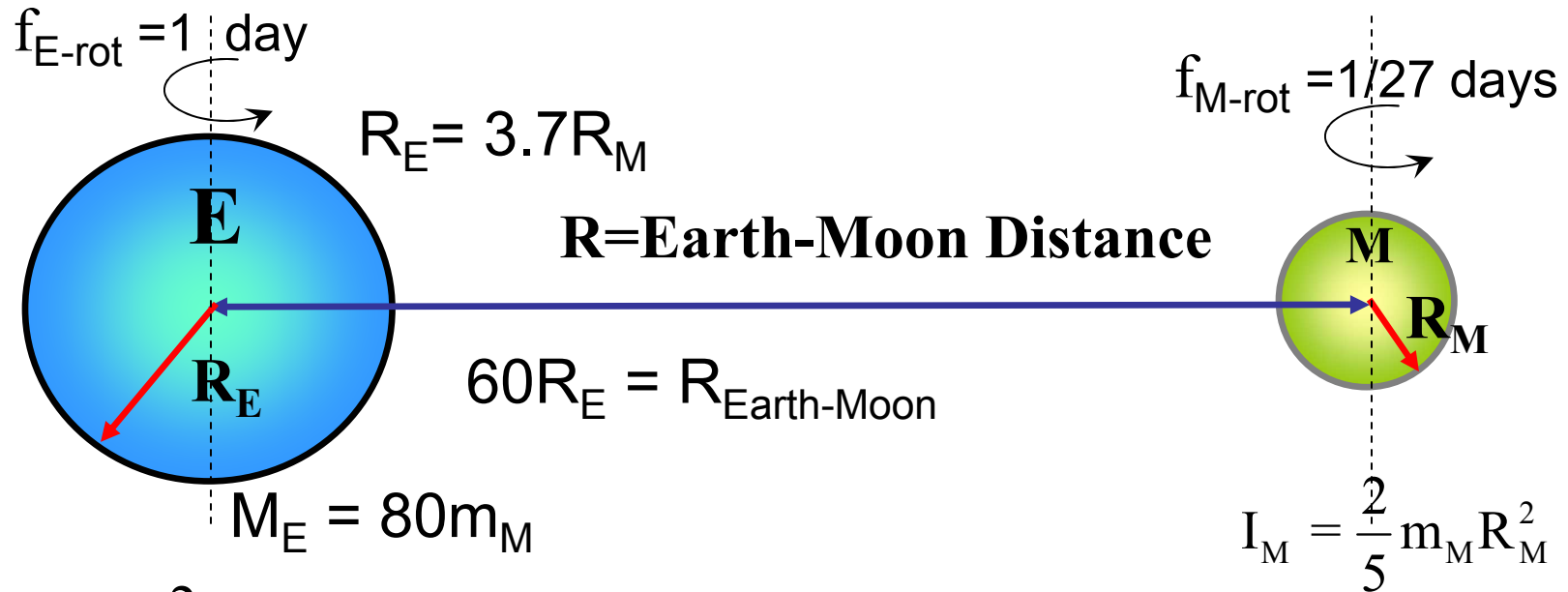
$$\tau = 27 (\text{now})$$

$$L_{M\text{-orbit}} = m_M R^2 \left(\frac{2\pi}{T_{M\text{-orbit}}} \right)$$

$$L_{M\text{-orbit}} = \frac{M_E}{80} (rR_E)^2 \left(\frac{2\pi}{\tau T_{E\text{-rot}}} \right)$$

$$L_{M\text{-orbit}} = \frac{r^2}{80\tau} \left[\frac{M_E R_E^2 2\pi}{T_E} \right]$$





$$I_E = \frac{2}{5}M_ER_E^2$$

$$I_M = \frac{2}{5}m_MR_M^2 = \frac{2}{5} \frac{M_E}{80} \left(\frac{R_E}{3.7}\right)^2 = (0.000913)I_E$$

$$L_{E-rot} = I_E (2\pi f_{E-rot})$$

$$L_{M-rot} = I_M (2\pi f_{M-rot})$$

$$L_{M-rot} = (0.000913)I_E (2\pi f_{E-rot}) \frac{1}{27}$$

$$L_{M-rot} = (0.0000338)L_{E-rot}$$

