

Velocity, acceleration, and force in rocket demo



$$t = 0$$

$$v_i = 0$$

$$x = 0$$

observation

$$\bar{v} = ? \quad \bar{v} = \frac{\Delta x}{\Delta t} = \frac{10\text{m}}{4\text{s}} = 2.5 \frac{\text{m}}{\text{s}}$$

$$t = 4\text{s}$$

$$x = 10\text{m}$$

one way

$$a = ? \quad (\text{assume const } a)$$

$$\Delta x = \frac{1}{2}at^2 \Rightarrow a = \frac{2(\Delta x)}{t^2}$$

$$a = \frac{2(10\text{m})}{(4\text{s})^2} = 1.25 \frac{\text{m}}{\text{s}^2}$$

$$v_f = ? \quad v_f = at = \left(1.25 \frac{\text{m}}{\text{s}^2}\right)(4\text{s}) = 5 \frac{\text{m}}{\text{s}}$$

another way

$$v_f = ? \quad (\text{assume } a = \text{const}) \quad \bar{v} = \frac{v_f + v_i}{2}$$

$$2.5 \frac{\text{m}}{\text{s}} = \frac{v_f + 0}{2} \Rightarrow v_f = 5 \frac{\text{m}}{\text{s}}$$

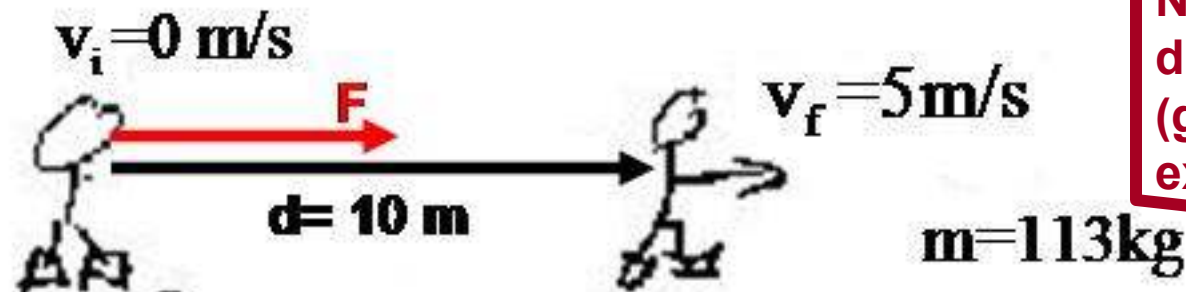
$$a = ? \quad a = \frac{v_f - v_i}{\Delta t}$$

$$a = \frac{(5-0)\frac{\text{m}}{\text{s}}}{4\text{s}} = 1.25 \frac{\text{m}}{\text{s}^2}$$

$$F = ? \quad F = ma$$

$$F = (113\text{kg})(1.25\text{m/s}^2) = 141 \text{ N (about 32 lb)}$$

Kinetic energy & Work Energy Theorem in rocket demo



Note Work done here is entirely due to non conservative forces (gas pressure release-like explosion).

$$W_{nc} = \Delta E = \Delta K$$

$$\Delta KE_{\text{Croft}} = \frac{1}{2}mv_f^2 - \frac{1}{2}mv_i^2$$

$$\Delta KE_{\text{Croft}} = \left\{ \frac{1}{2}(113)5^2 - 0 \right\} \text{kg} \frac{\text{m}^2}{\text{s}^2} = 1412 \text{ J}$$

Work Energy
Theorem

$$W = \Delta KE$$

Work

$$W = F \cdot d$$

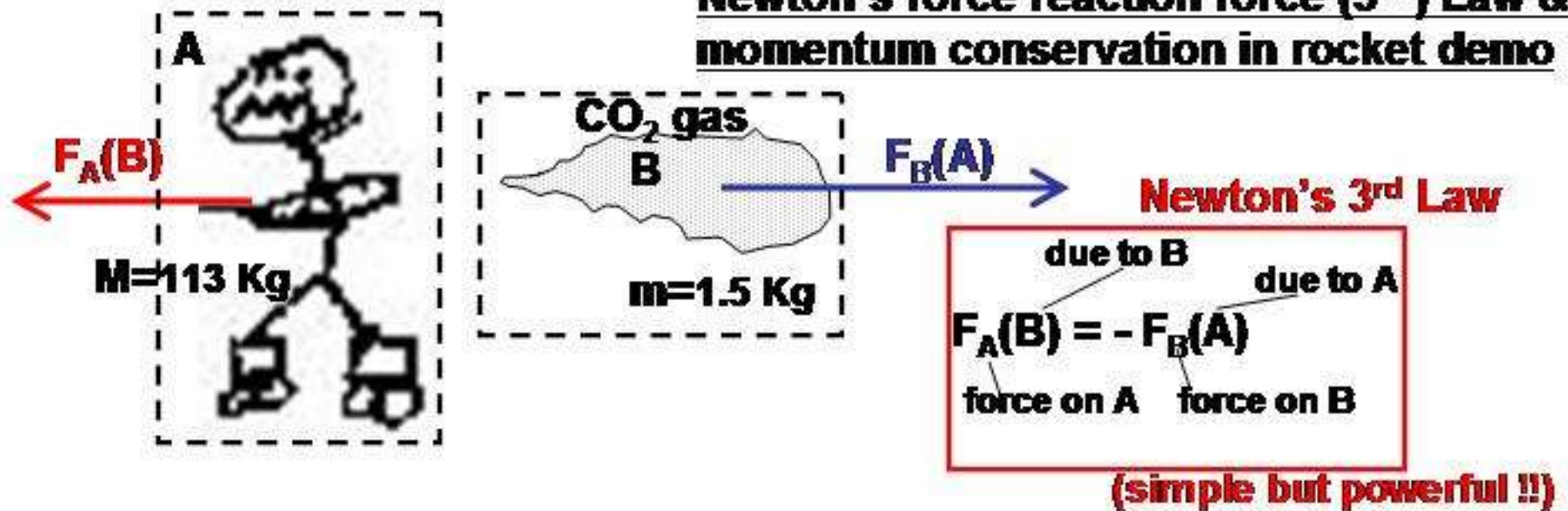
$$\therefore F \cdot d = \Delta KE = \frac{1}{2}mv_f^2 - \frac{1}{2}mv_i^2$$

$$F(10\text{m}) = \frac{1}{2}(113)5^2 = 1412 \text{ J}$$

$$F = \frac{1412 \text{ J}}{10 \text{ m}} = 141 \frac{\text{J}}{\text{m}} = 141 \text{ N}$$

force of gas on croft

Newton's force reaction force (3rd) Law & momentum conservation in rocket demo



Newton's 2nd Law $F = ma$ or $F = \frac{\Delta p}{\Delta t}$ where $p = mv$

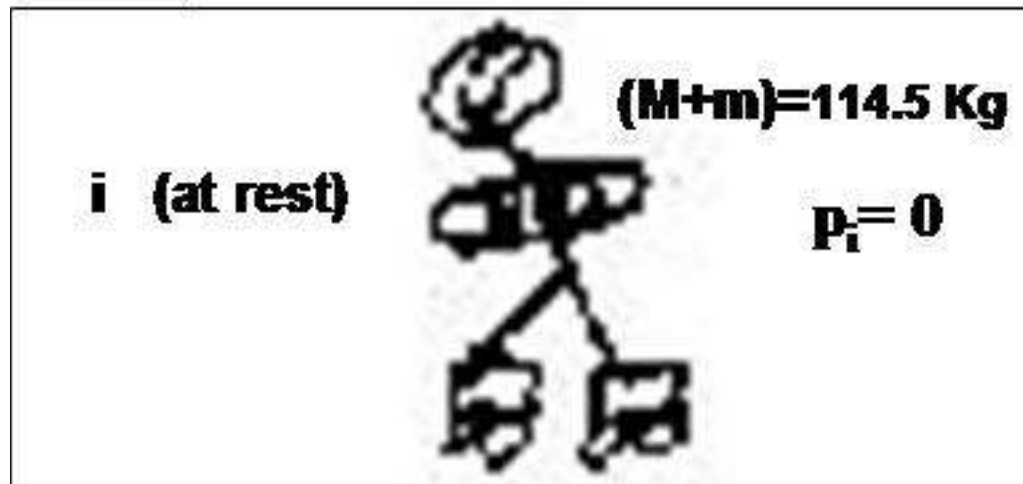
A+B system: no external forces \Rightarrow momentum conservation

$$F_{\text{tot}} = F_A + F_B = 0 \Rightarrow \frac{\Delta p_{\text{tot}}}{\Delta t} = 0 \Rightarrow \Delta p_{\text{tot}} = \Delta p_A + \Delta p_B = 0$$

\Downarrow

$$\Delta p_A = -\Delta p_B$$

Conservation of momentum in rocket demo



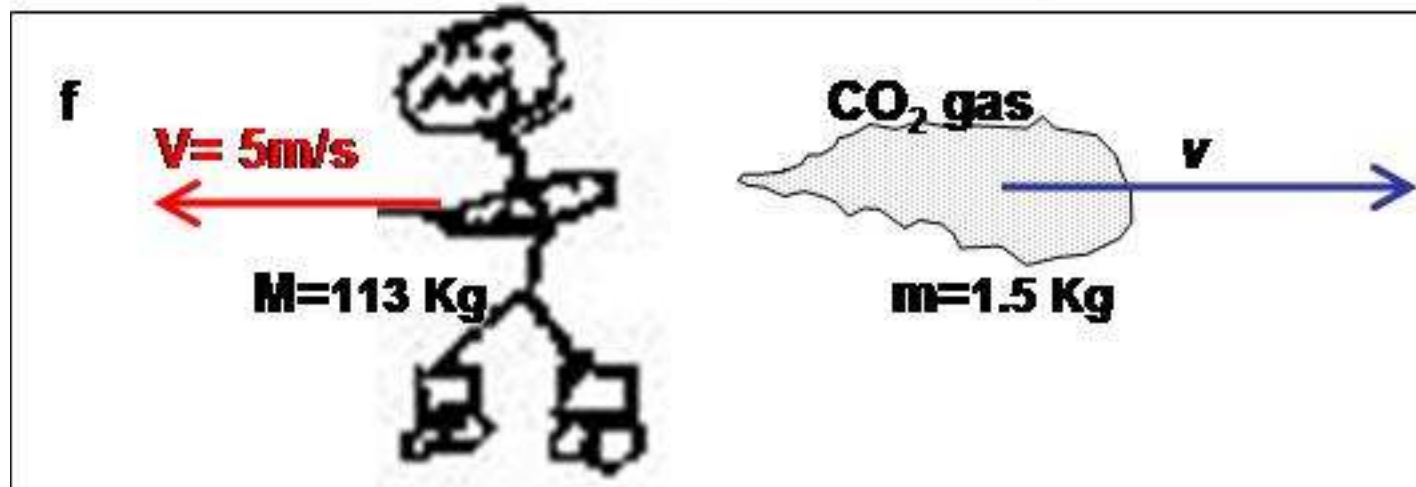
initial momentum \rightarrow final momentum

$P_i = P_f$

$$0 = +MV - m v$$

$$m v = MV$$

$$v = \frac{113 \text{ Kg}}{1.5 \text{ Kg}} 5 \frac{\text{m}}{\text{s}}$$



$$v = 75 \frac{\text{m}}{\text{s}}$$

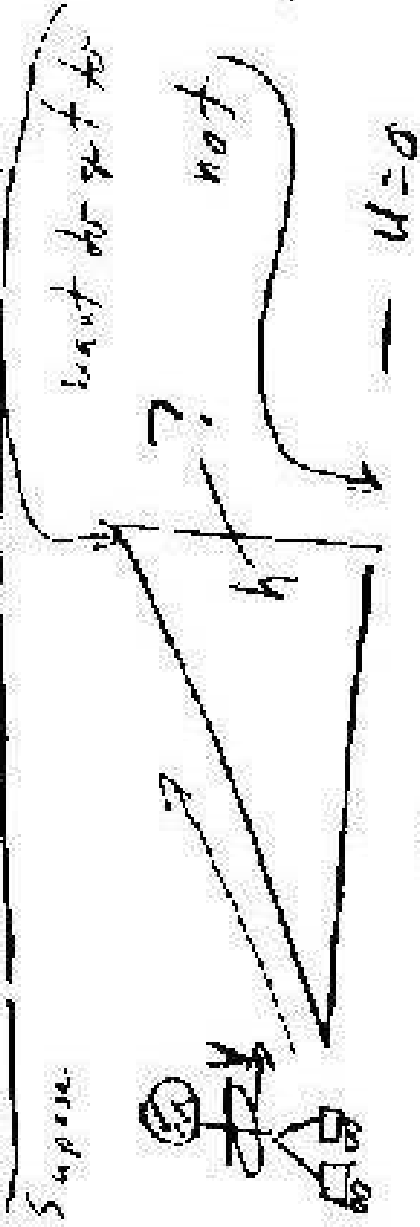
for CO_2 gas

$$m = 1.5 \text{ Kg} = \text{mass tank after} - \text{before} = (16.5 - 15) \text{ Kg}$$

kinetic to potential energy conversion:
design of stop-ramp for rocket demo

Build inclined plane to stop craft

Suppose



① how high could h be

$$E_i = \frac{1}{2} m v_i^2 + 0 \quad E_f = 0 + mgh \quad v_f = 0 \text{ not}$$

$$E_i = E_f$$

$$\frac{1}{2} m v_i^2 = mgh$$

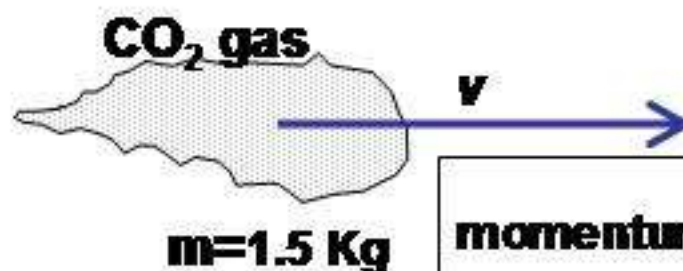
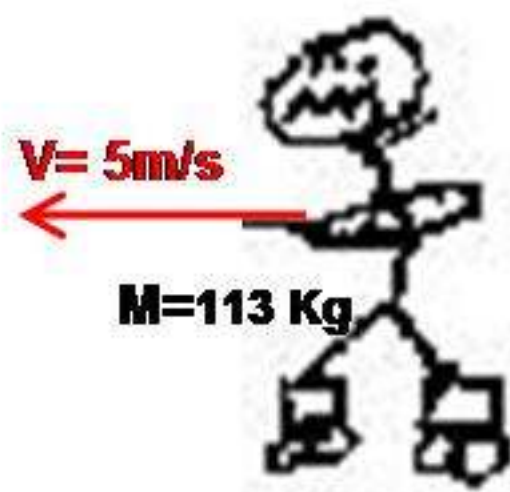
$$h = \frac{\frac{1}{2} v_i^2}{g} = \frac{1}{2} \frac{5^2}{9.8} \approx 1.25 \text{ m}$$

ramp 1.25 m high should

stop craft

P.S. physics works but not a very good idea

Division of energy & momentum in rocket demo



momentum conservation
 $MV = m v \Rightarrow \frac{v}{V} = \frac{M}{m}$

$$\frac{KE_{\text{gas}}}{KE_{\text{perf}}} = \frac{\frac{1}{2} m v^2}{\frac{1}{2} M V^2} = \frac{m}{M} \left(\frac{v}{V} \right)^2 = \frac{m}{M} \left(\frac{M}{m} \right)^2 = \frac{M}{m}$$

$$\frac{KE_{\text{gas}}}{KE_{\text{perf}}} = \frac{M}{m} = \frac{113}{1.5} = 75!!$$

Energy to gas!!

$\frac{p_m}{p_M} = \frac{m v}{M V} = 1$ in general
momentum equally to both !!

$$\frac{KE_m}{KE_M} = \frac{M}{m}$$

Energy goes to the little guy!!

**In radioactive decay
most energy goes to
smaller particle**

