

### HW#3 solutions

① Start with the Lagrangian:

$$\mathcal{L} = \frac{m}{2} (\dot{r}^2 + r^2 \dot{\theta}^2) + k \frac{e^{-ar}}{r} \quad (*)$$

EoM for  $\theta$  leads to  $l = mr^2 \dot{\theta} = \text{const.}$

EoM for  $r$ :

$$\begin{cases} \frac{\partial \mathcal{L}}{\partial \dot{r}} = m \dot{r}, \\ \frac{\partial \mathcal{L}}{\partial r} = mr \dot{\theta}^2 - k(1+ar) \frac{e^{-ar}}{r^2} \end{cases}$$

$$\ddot{r} = \underbrace{r \dot{\theta}^2}_{\frac{l^2}{m^2 r^4}} - \frac{k}{m} (1+ar) \frac{e^{-ar}}{r^2} \quad (\ominus)$$

$$\ominus \frac{l^2}{m^2 r^3} - \frac{k}{m} (1+ar) \frac{e^{-ar}}{r^2} \quad (**)$$

Note that the effective potential

$$(*) \rightarrow V'(r) = -k \frac{e^{-ar}}{r} + \frac{l^2}{2mr^2}$$
$$\left. \frac{dV'(r)}{dr} \right|_{r_0} = 0 \text{ leads to}$$

$$k e^{-ar_0} \left( \frac{1}{r_0} + a \right) - \frac{l^2}{mr_0^2} = 0,$$

which is satisfied by  $r_0 = \infty$  and

$$\text{by } \frac{km}{\ell^2} (1+ar_0) r_0 = \ell^2 ar_0$$

↑

transcendental eq'n for  $r_0$

Note that if  $a=0$ ,

$$\frac{km}{\ell^2} r_0 = 1 \Rightarrow r_0 = \frac{\ell^2}{mk} \text{ as expected.}$$

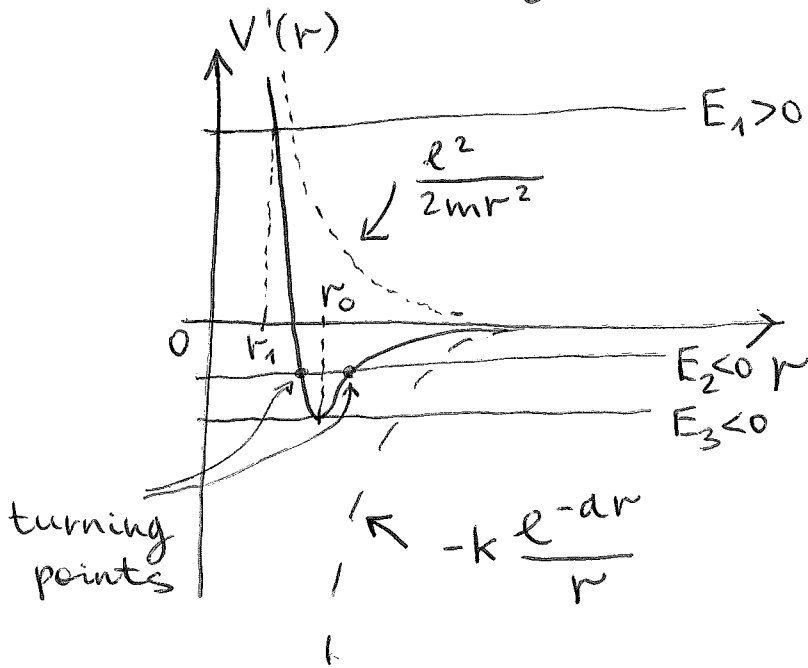
$$\text{Note also that } V'(r_0) = \frac{\ell^2}{2mr_0^2} \frac{ar_0-1}{ar_0+1}$$

Finally, Eq. (\*\*) gives

$$\ddot{r} \Big|_{r_0} = \frac{\ell^2}{m^2 r_0^3} - \frac{k}{m} (1+ar_0) \frac{1}{r_0^2} \frac{\ell^2}{km} \frac{1}{(1+ar_0)r_0} =$$

$= 0$ , as expected for a  
circular orbit.

Graphically,



Clearly, a particle with  $E_1 > 0$  has an unbounded trajectory: it comes in, turns around at  $r = r_1$ , ~~and~~ and goes back out. The  $r < r_1$  region is forbidden. For  $E_2 < 0$ , the particle's orbit will be bounded between the two turning points shown in the Fig. above, but not necessarily closed. Finally, at  $E_3 = V'(r_0)$ , the particle will be in a circular orbit.

2.

3.11

Gravitational forces:

$$U(r) = -\frac{k}{r}, \text{ leading to}$$

$$\mathcal{L} = \frac{\mu}{2} (\dot{r}^2 + r^2 \dot{\theta}^2) + \frac{k}{r}, \text{ where}$$

$$\mu = \frac{m_1 m_2}{m_1 + m_2} \text{ is the reduced mass.}$$

EOM is given by:

$$\mu \ddot{r} = \mu r \dot{\theta}^2 - \frac{k}{r^2}. \quad (*)$$

For circular orbits,  $\ddot{r}|_{r_0} = 0$ , s.t.

$$r_0^3 = \frac{k}{\mu \dot{\theta}^2} = \frac{k \tau^2}{4\pi^2 \mu}$$

$$\dot{\theta} = \frac{2\pi}{\tau} = \text{const for circular orbits}$$

When the particles are stopped,

$\dot{\theta} = 0$  in Eq. (\*), yielding

$$2\dot{r} \times \left| \ddot{r} = -\frac{k}{\mu r^2} \right. \Leftrightarrow \text{need to get } r(t) \\ \text{with } \begin{cases} r(0) = r_0, \\ \dot{r}(0) = 0 \end{cases}$$

$$2\dot{r}\ddot{r} = -\frac{2k}{\mu r^2}\dot{r}, \text{ or}$$

$$\frac{d}{dt}(\dot{r}^2) = \frac{d}{dt}\left(\frac{2k}{\mu r}\right) \Rightarrow \dot{r}^2 = \frac{2k}{\mu r} + C$$

$$\dot{r}(0) = 0 \Rightarrow C = -\frac{2k}{\mu r_0}$$

$$\text{Then } \frac{dr}{dt} = \sqrt{\frac{2k}{\mu}} \sqrt{\frac{1}{r} - \frac{1}{r_0}}$$

Now, consider

$$\Delta t = \int_{r_0}^0 dr \left(\frac{dt}{dr}\right) = \sqrt{\frac{\mu}{2k}} \int_{r_0}^0 dr \sqrt{\frac{rr_0}{r_0-r}}$$

$$\int_{r_0}^0 dr \sqrt{\frac{rr_0}{r_0-r}} = r_0 \int_1^0 du \sqrt{\frac{ur_0}{1-u}} = r_0^{3/2} \int_1^0 du \sqrt{\frac{u}{1-u}} \quad \text{⊖}$$

$$\begin{cases} u = \frac{r}{r_0} \\ du = \frac{dr}{r_0} \end{cases}$$

$$\begin{cases} u = \sin^2 x \\ du = 2\sin x \cos x dx \end{cases}$$

$$\text{⊖ } r_0^{3/2} \int_{\frac{\pi}{2}}^0 dx \times 2\sin x \cos x \frac{\sin x}{\cos x} = 2r_0^{3/2} \int_{\frac{\pi}{2}}^0 dx \sin^2 x =$$

$$= 2r_0^{3/2} \left[ \frac{\pi}{4} - \frac{1}{2} \sin(2x) \right]_{\frac{\pi}{2}}^0 = 2r_0^{3/2} \frac{\pi}{4}$$

Finally,

$$\Delta t = \sqrt{\frac{\mu}{2k}} r_0^{3/2} \frac{\pi}{2} = \sqrt{\frac{\mu}{2k}} \sqrt{\frac{k r_0^2}{4 \mu^2 \mu}} \frac{\pi}{2} = \frac{\pi}{4\sqrt{2}},$$

as desired



$$\Leftrightarrow \frac{l^2}{mk} \frac{\sin \theta}{(1 + \cos \theta)^2} \dot{\theta} = r \dot{\theta} \frac{\sin \theta}{1 + \cos \theta}$$

Then  $v_p^2 = r^2 \dot{\theta}^2 \left[ \frac{\sin^2 \theta}{(1 + \cos \theta)^2} + 1 \right] =$

$$= r^2 \dot{\theta}^2 \frac{2 + 2 \cos \theta}{(1 + \cos \theta)^2} = \frac{2r^2 \dot{\theta}^2}{1 + \cos \theta} = \frac{2r^2 l^2}{m^2 r^4} \frac{mk}{l^2} r \Leftrightarrow$$

$$\begin{cases} \dot{\theta}^2 = \frac{l^2}{m^2 r^4}, \\ \frac{1}{1 + \cos \theta} = \frac{mk}{l^2} r \end{cases} \nearrow$$

$$\Leftrightarrow \frac{2k}{mr}$$

Finally,  $v_p = \sqrt{2} \sqrt{\frac{k}{mr}} = \underline{\underline{\sqrt{2} v_c}}$



4. 3.20

(a) Circular orbit:

$$T = \frac{2\pi}{\dot{\theta}} \quad \& \quad l = m r^2 \dot{\theta} \Rightarrow T = \frac{2\pi m r^2}{l}$$

Here,  $f(r) = -\frac{k}{r^2} - m C r$

$$|f(r)| = m \frac{v_c^2}{r} = m r \dot{\theta}^2 = m r \frac{l^2}{m^2 r^4} = \frac{l^2}{m r^3}$$

$v_c = r \dot{\theta}$

So,  $\frac{k}{r_0^2} + m C r_0 = \frac{l^2}{m r_0^3}$ , or

$$l = \sqrt{m r_0 k + m^2 C r_0^4}$$

Finally,  $T = \frac{2\pi m r_0^2}{\sqrt{m r_0 k + m^2 C r_0^4}} =$

$$= \frac{2\pi}{\sqrt{C + \frac{k}{m r_0^3}}}$$

Note that  $\omega = \frac{2\pi}{T} = \sqrt{C + \frac{k}{m r_0^3}}$

(b) Use

$$u = u_0 + d \cos(\beta \theta) \quad (3.45)$$

$\underbrace{\quad}_{\text{"1/r"}}$

and

$$\beta^2 = 3 + \frac{r}{f} \frac{df}{dr} \Big|_{r=r_0} \quad (3.46) \quad \text{for a circular orbit of radius } r_0$$

Here,  $f = -\frac{k}{r^2} - mCr$ , which gives

$$\frac{df}{dr} = \frac{2k}{r^3} - mC$$

$$\text{Then } \beta^2 = 3 + \frac{r_0}{\frac{k}{r_0^2} + mCr_0} \left( mC - \frac{2k}{r_0^3} \right) =$$

$$= \frac{\frac{3k}{r_0^2} + 3mCr_0 + mCr_0 - \frac{2k}{r_0^2}}{\frac{k}{r_0^2} + mCr_0} = \frac{4C + \frac{k}{mr_0^3}}{C + \frac{k}{mr_0^3}}$$

From (3.45) it is clear that

$$T_{osc} = \frac{T}{\beta} = \frac{2\pi / \sqrt{C + \frac{k}{mr_0^3}}}{\frac{\sqrt{4C + \frac{k}{mr_0^3}}}{\sqrt{C + \frac{k}{mr_0^3}}}} = \frac{2\pi}{\sqrt{4C + \frac{k}{mr_0^3}}}$$

use T from (a)

$$\text{Note that } \omega_{osc} = \frac{2\pi}{T_{osc}} = \sqrt{4C + \frac{k}{mr_0^3}}$$